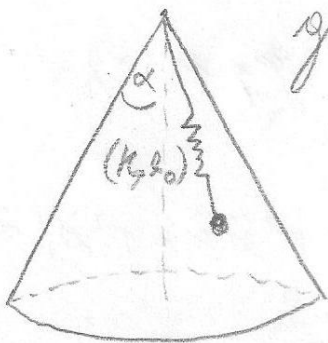


Aux 9

P1



g ↓

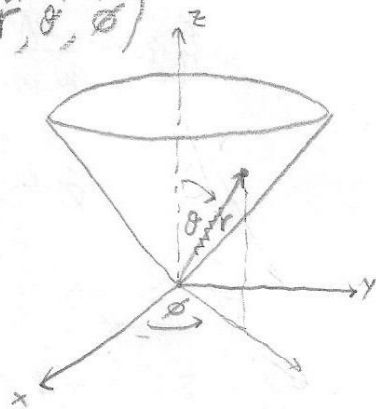
¿Coordenadas?

- Rotacional es mucho más complicado.
 - Cilíndrico es difícil porque varía la altura, el radio polo y el ángulo polo.
 - Esférico es malo porque no sabemos la trayectoria
- ⇒ Esféricas! $(\hat{r}, \hat{\theta}, \hat{\phi})$

a) • En esféricas con origen en el vértice del cono:

⇒ r variable, ϕ variable, θ constante (cono)

⇒ $\theta = \alpha \Rightarrow \dot{\theta} = \ddot{\theta} = 0$



• Condiciones iniciales: $\vec{r}(0) = a \hat{r}$

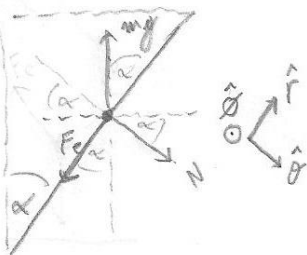
$$\vec{v}(0) = \dot{r}(0) \hat{r} + r \dot{\phi} \hat{\phi} + r \dot{\theta} \hat{\theta} + r(0) \dot{\phi}(0) \sin \alpha \hat{\phi} = v_0 \hat{\phi}$$

$$\Rightarrow \dot{r}(0) = 0, \quad \dot{\phi}(0) = \frac{v_0}{a \sin \alpha}, \quad r(0) = a \quad (1)$$

Ec. de torque: $\frac{d\vec{L}}{dt} = \sum \vec{\tau}$ con $\vec{L} = m \vec{r} \times \vec{v}$

DCL

$$\Rightarrow \sum \vec{\tau} = \vec{\tau}_N + \vec{\tau}_{mg} + \vec{\tau}_{F_c} \quad (\vec{\tau} = \vec{r} \times \vec{F})$$



$$\vec{\tau}_{F_c} = r \hat{r} \times k(l_0 - r) \hat{r} = 0$$

$$\vec{\tau}_N = r N (\hat{r} \times \hat{\theta}) = -r N \hat{\phi}$$

$$\vec{\tau}_{mg} = r \hat{r} \times (mg \cos \alpha \hat{r} - mg \sin \alpha \hat{\theta}) = -r mg \sin \alpha \hat{\phi}$$

momento angular: $\vec{L} = m \vec{r} \times \vec{v} = m r \hat{r} \times (\dot{r} \hat{r} + r \dot{\phi} \hat{\phi} + r \sin \alpha \dot{\theta} \hat{\theta})$

$\Rightarrow \vec{L} = -m r^2 \dot{\phi} \sin \alpha \hat{\theta}$

Ec. de Torque: $\frac{d}{dt} (-m r^2 \dot{\phi} \sin \alpha \hat{\theta}) = r \hat{\phi} (N - m g \sin \alpha) \quad (2)$

$-m \sin \alpha \frac{d}{dt} (r^2 \dot{\phi} \hat{\theta}) = -m \sin \alpha (2 r \dot{r} \dot{\phi} \hat{\theta} + r^2 \ddot{\phi} \hat{\theta} + r^2 \dot{\phi} [\dot{\phi} \cos \theta \hat{\theta} - \dot{\theta} \hat{r}])$

$r \neq 0 \Rightarrow$
 $(2) \quad \left(g - \frac{N}{m \sin \alpha} \right) \hat{\phi} = r \dot{\phi}^2 \cos \theta \hat{\theta} + (2 \dot{r} \dot{\phi} + r \ddot{\phi}) \hat{\theta}$
 $-m r \sin \alpha$

$\Rightarrow \left[r \dot{\phi}^2 \cos \alpha = g - \frac{N}{m \sin \alpha} \right] \quad (3)$

$\Rightarrow 2 \dot{r} \dot{\phi} + r \ddot{\phi} = 0 \quad \cdot r \Rightarrow 2 r \dot{r} \dot{\phi} + r^2 \ddot{\phi} = \frac{d}{dt} (r^2 \dot{\phi}) = 0$

$\Rightarrow \left[r^2 \dot{\phi} \text{ es cantidad conservada, por C.I tenemos:} \right]$

b) $r^2 \dot{\phi} = r(0) \dot{\phi}(0) \stackrel{(1)}{=} a^2 \frac{v_0}{a \sin \alpha} = \frac{a v_0}{\sin \alpha} \Rightarrow \dot{\phi} = \frac{a v_0}{r^2 \sin \alpha} \quad (4)$

Ec. de mov en \hat{r} : $\hat{r} \mid m(\ddot{r} - r \dot{\phi}^2 - r \dot{\phi}^2 \sin^2 \alpha) = m g \cos \alpha + K(l_0 - r)$

$\frac{1}{m} \Rightarrow \left[\ddot{r} = \frac{a^2 v_0^2}{r^3} + g \cos \alpha + \frac{K}{m} (l_0 - r) \right] \quad \text{ec. diferencial para } r. \quad (5)$

c) con el resultado: $\ddot{r} = \dot{r} \frac{d\dot{r}}{dr}$ podemos resolver la EDO y tener $\dot{r}(r)$.



(5) =>

$$\dot{r} \frac{dr}{dr} = \frac{a^2 v_0^2}{r^3} + g \cos \alpha + \frac{k}{m} (l_0 - r)$$

$$v_f = v_i \Rightarrow \dot{r}_i = \dot{r}_f = 0$$

$$r_f = \frac{a}{2}$$

$$\Rightarrow \int_0^a \dot{r} dr = \int_a^{\frac{a}{2}} \frac{a^2 v_0^2}{r^3} dr + \int_a^{\frac{a}{2}} g \cos \alpha dr + \int_a^{\frac{a}{2}} \frac{k}{m} l_0 dr - \int_a^{\frac{a}{2}} \frac{k}{m} r dr$$

$$\Rightarrow 0 = -2a^2 v_0^2 \left(\frac{4}{a^2} - \frac{1}{a^2} \right) + \left(g \cos \alpha + \frac{k l_0}{m} \right) \left(\frac{a}{2} - a \right) - \frac{k}{2m} \left(\left(\frac{a}{2} \right)^2 - a^2 \right)$$

$$\Rightarrow 0 = -6v_0^2 + \frac{a}{2} \left(g \cos \alpha + \frac{k l_0}{m} \right) - \frac{k a^2}{2m} \left(\frac{1}{4} - 1 \right)$$

$$\Rightarrow \boxed{\frac{3ka^2}{8m} - \frac{a}{2} \left(g \cos \alpha + \frac{k l_0}{m} \right) - 6v_0^2 = 0}$$

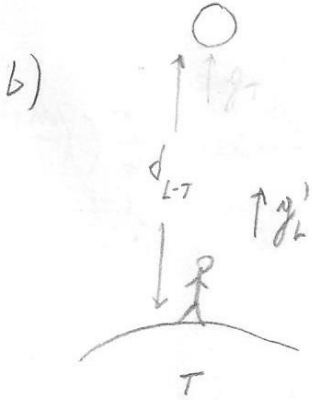
for the relationship for parameter $(k, a, m, \alpha, v_0, l_0, g)$

P2

a) aceleración en la Tierra: $g_T = \frac{GM_T}{R_T^2} \Rightarrow G = \frac{g_T R_T^2}{M_T} \quad (1)$

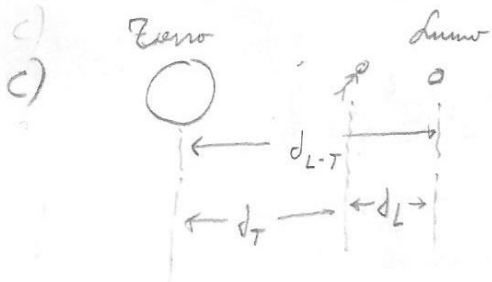
aceleración en la luna: $g_L = \frac{GM_L}{R_L^2} \Rightarrow g_L = g_T \frac{M_L}{M_T} \left(\frac{R_T}{R_L}\right)^2 \approx 10 \cdot 10^{-2} (4)^2 \approx 1.6 \frac{m}{s^2}$

$\Rightarrow g_L = 1.6 \frac{m}{s^2}$



$$g_L' = \frac{GM_L}{d_{L-T}^2} = g_T \frac{M_L}{M_T} \left(\frac{R_T}{d_{L-T}}\right)^2 \approx 10 \cdot 10^{-2} \left(\frac{64 \cdot 10^2}{38 \cdot 10^4}\right)^2$$

$$\approx 10^{-1} (1.6 \cdot 10^{-2})^2 = 2.56 \cdot 10^{-5} \Rightarrow g_L' = 2.56 \cdot 10^{-5} \frac{m}{s^2}$$



Por ser centro de masas preceded: $g_L' = g_T'$

$$\Rightarrow \frac{GM_T}{d_T^2} = \frac{GM_L}{d_L^2} \Rightarrow \frac{M_T}{M_L} = \left(\frac{d_T}{d_L}\right)^2 \approx 10^2$$

$$\Rightarrow d_T = 10 d_L \quad y \quad d_T + d_L = d_{L-T} = 3.8 \cdot 10^5 \text{ km}$$

$$\Rightarrow d_L \cdot 11 = 3.8 \cdot 10^5 \text{ km} \Rightarrow d_L \approx 3.4 \cdot 10^4 \text{ km}$$

$$\Rightarrow d_T = 3.46 \cdot 10^5 \text{ km}$$