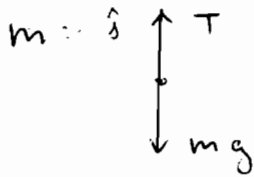
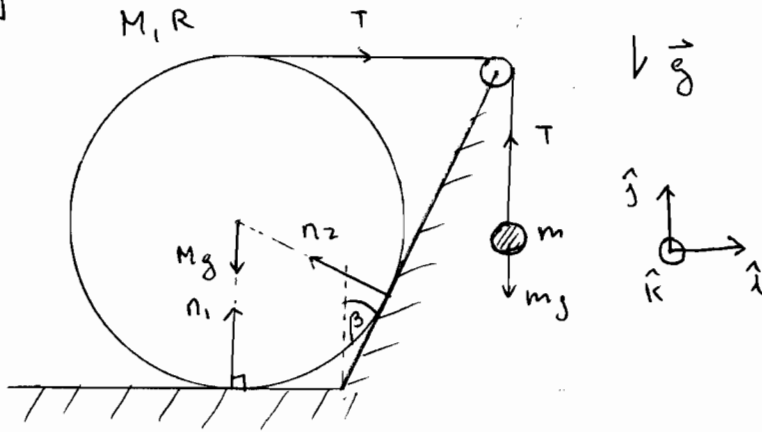


P1]

Datos: M, R

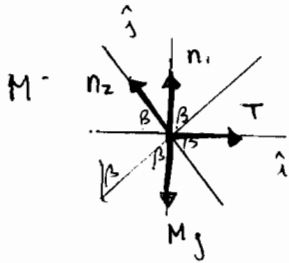
$i, m?$



$$\vec{T} + m\vec{g} = m\vec{a}_m$$

$$\hat{j}) \quad T - mg = m a_m \quad (1)$$

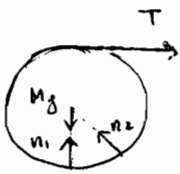
$\rightarrow 0$ por enunciado



$$\vec{T} + \vec{n}_1 + \vec{n}_2 + M\vec{g} = M\vec{a}_{cm}$$

$$\hat{i}) \quad T + 0 - n_2 \cos \beta + 0 = 0 \quad (2)$$

$$\hat{j}) \quad 0 + n_1 + n_2 \sin \beta - Mg = 0 \quad (3)$$



$$\vec{\tau}_{cm}(\vec{T}) + \vec{\tau}_{cm}(M\vec{g}) + \vec{\tau}_{cm}(\vec{n}_1) + \vec{\tau}_{cm}(\vec{n}_2) = I_{cm} \vec{\alpha}$$

$$\hat{k}) \quad -RT = \frac{1}{2} MR^2 \alpha \quad \text{pero } a_m = R \cdot \alpha$$

$$\Rightarrow -RT = \frac{1}{2} MR^2 \frac{a_m}{R}$$

$$\Rightarrow a_m = -\frac{2T}{M} \quad (4)$$

Condición: $n_1 = 0$

$$\left. \begin{array}{l} (3) \rightarrow n_2 \sin \beta = Mg \\ (2) \rightarrow n_2 \cos \beta = T \end{array} \right\} \begin{array}{l} (3)/(2) \Rightarrow \tan \beta = \frac{Mg}{T} \\ \Rightarrow T = \frac{Mg}{\tan \beta} \quad (5) \end{array}$$

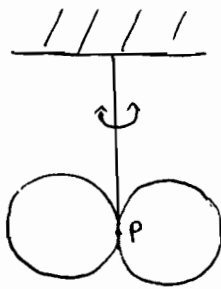
$$(4) \text{ en } (1) \Rightarrow T - mg = -\frac{2m}{M} T$$

$$\Rightarrow T = \frac{m}{M+2m} Mg \quad (6)$$

$$(5) = (6) \Rightarrow \frac{Mg}{\tan \beta} = \frac{m}{M+2m} Mg \Rightarrow$$

$$\boxed{m = \frac{M}{\tan \beta - 2}}$$

P2]



Datos: $M, R, z = -\lambda \theta$

$$(a) \quad I_{\text{est}} = \frac{2}{5} MR^2$$

gr CM

Steiner $I_P = I_{\text{cm}} + MD^2$

$$I_{\text{est}} = \frac{2}{5} MR^2 + MR^2 = \frac{7}{5} MR^2$$

gr P

$$\Rightarrow \boxed{I_{\text{est}} = \frac{14}{5} MR^2}$$

gr P

$$(b) \quad \tau_P = I_P \ddot{\theta}$$

$$\Rightarrow -\lambda \theta = \frac{14}{5} MR^2 \ddot{\theta}$$

$$\Rightarrow \ddot{\theta} + \underbrace{\frac{5\lambda}{14MR^2}}_{\omega^2} \theta = 0$$

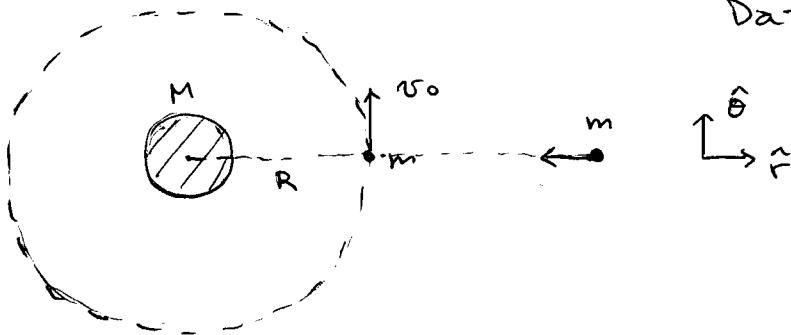
$$\Rightarrow \boxed{T = 2\pi \sqrt{\frac{14MR^2}{5\lambda}}}$$

$$(c) \quad \theta(t) = A \cos\left(\sqrt{\frac{5\lambda}{14MR^2}} t + \phi\right)$$

$$\text{CI: } \left. \begin{array}{l} \theta(0) = \theta_0 \\ \dot{\theta}(0) = 0 \end{array} \right\} \Rightarrow \begin{array}{l} A = \theta_0 \\ \phi = 0 \end{array}$$

$$\Rightarrow \boxed{\theta(t) = \theta_0 \cos\left(\sqrt{\frac{5\lambda}{14MR^2}} t\right)}$$

P3]

Datos: M, m, R, v_1 

$$(a) \quad \vec{F} = m \vec{a}$$

$$\Rightarrow -\frac{GMm}{R^2} \hat{r} = -m \frac{v_0^2}{R} \hat{r}$$

$$\Rightarrow \boxed{v_0 = \sqrt{\frac{GM}{R}}}$$

$$(b) \quad \vec{P}_f - \vec{P}_i = \vec{F} \cdot \Delta t$$

$$\hat{\theta}) (2m) v_\theta - m v_0 = \underbrace{F_\theta}_0 \cdot \Delta t$$

$$\Rightarrow v_\theta = \frac{v_0}{2} = \frac{1}{2} \sqrt{\frac{GM}{R}}$$

$$\hat{r}) (2m) v_r - (-m v_1) = \underbrace{F_r}_0 \cdot \Delta t \quad \text{se asume impacto instantáneo}$$

$$\Rightarrow v_r = -\frac{v_1}{2}$$

$$\text{Entonces } \vec{V} = -\frac{v_1}{2} \hat{r} + \frac{1}{2} \sqrt{\frac{GM}{R}} \hat{\theta}$$

(c) Energía después del choque

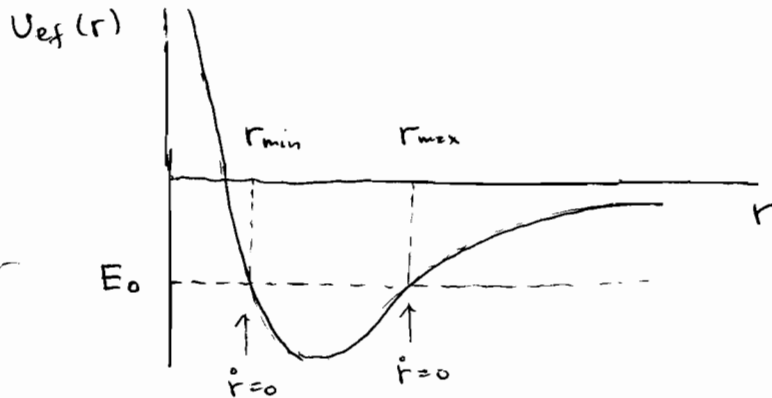
$$E_o = \frac{1}{2} (2m) \vec{V}^2 - \frac{GM(2m)}{R}$$

$$= \frac{1}{2} (2m) \left[\frac{v_1^2}{4} \underbrace{\hat{r} \cdot \hat{r}}_1 + -\left(\frac{v_1}{2}\right) \left(\frac{1}{2} \sqrt{\frac{GM}{R}}\right) \underbrace{\hat{r} \cdot \hat{\theta}}_0 + \frac{1}{4} \frac{GM}{R} \underbrace{\hat{\theta} \cdot \hat{\theta}}_1 \right] - \frac{2GMm}{R}$$

$$= \frac{1}{4} m v_1^2 - \frac{7}{4} \frac{GMm}{R} \quad (1)$$

La energía mecánica se puede expresar en función de un potencial efectivo:

$$E = \frac{1}{2} m \dot{r}^2 + \underbrace{\frac{L_0^2}{2mr^2} - \frac{GMm}{r}}_{U_{\text{ef}}(r)} \quad (2)$$



$$(1) = (2) \Rightarrow E_0 = \frac{1}{2} m \dot{r}^2 + \frac{L_0^2}{2mr^2} - \frac{GMm}{r}$$

0 (se impone para encontrar r_{\min} y r_{\max})

\Rightarrow Se resuelve la ecuación cuadrática y

Se obtiene r_{\min} (perigeo) y r_{\max} (apogeo)

(d) Para que el cuerpo proyectil-satélite escape del campo gravitacional del planeta $\Rightarrow E_0 = 0$

(trayectoria parabólica)

$$\text{de (1)} \Rightarrow \frac{1}{2} m v_1^2 - \frac{GMm}{R} = 0$$

$$\Rightarrow \boxed{v_1 = \sqrt{\frac{2GM}{R}}}$$